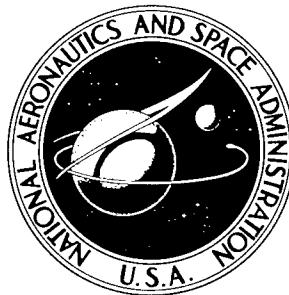


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OPTIMIZATION OF TIME-TEMPERATURE
PARAMETERS FOR CREEP AND
STRESS RUPTURE, WITH APPLICATION
TO DATA FROM GERMAN COOPERATIVE
LONG-TIME CREEP PROGRAM

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SUMMARY

By the use of orthogonal polynomials developed for discrete sets of data, the least-squares equations for determining the optimized stress-rupture parametric constants are obtained in nearly uncoupled form; thus the use of high-degree polynomials is permitted without the loss of significant figures. Optimum values of the constants can thereby be accurately obtained. The method is applied to the data obtained from the German cooperative long-time creep program by using a general parameter of which the Manson-Haferd and Larson-Miller parameters are special cases. Good correlation was obtained. An analysis is also made of creep data obtained for columbium alloy FS-85 with good results. A complete Fortran IV computer program is included to aid those wishing to use the method.

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INTRODUCTION

One method of extrapolating short-time creep-rupture data to predict long-time life involves the use of a time-temperature parameter. This concept is based on the assumption that all creep-rupture data for a given material can be correlated to produce a single "master curve" wherein the stress (or log stress) is plotted against a parameter involving a combination of time and temperature. Extrapolation to long times can then be obtained from this master curve, which can presumably be constructed by using only short-time data. Three well-known parametric methods are the Larson-Miller, Manson-Haferd, and Dorn parameters (refs. 1 to 3). These parametric methods have the great advantage, at least in theory, of requiring only a relatively small amount of data to establish the required master curve.

AII
More recently a general creep-rupture parameter was introduced by one of the authors (ref. 4) that includes most of the currently used parameters as special cases. The analysis in the present paper is therefore based on this general parameter.

A significant advance in the practical application of the parametric methods was the development of an objective least-squares method for determining the optimum values of the parametric constants without plotting and cross-plotting the data and without the use of judgment on the part of the analyst (ref. 5). This least-squares method involves, however, several practical difficulties that arise from the fact that in fitting the master curve by a polynomial, the set of linear algebraic equations for the coefficients (the normal equations) are very ill-conditioned. The determinant of these equations can be shown to be related to the Hilbert determinant (ref. 6), which rapidly approaches zero as its order increases. Thus for polynomials above the second degree, it is necessary to use double-precision arithmetic (16 significant digits or more) on the computer, and for the fifth degree and above the results become uncertain even with double-precision arithmetic. This difficulty is inherent in the normal least-squares equations and is not limited only to the stress-rupture problem.

The present report presents a method for avoiding the above difficulty by using orthogonal polynomials in the representation of the master curve (appendix A). The use of orthogonal polynomials for representing discrete sets of unequally spaced data is described in reference 6 and in more detail in reference 7. A further improvement can be obtained by performing a linear transformation on the stresses (or the logs of the stresses) so that all the values of stress (or log stress) lie between 2 and -2, as recommended in reference 7. As a result of these innovations, it became possible to perform all the computations in single-precision arithmetic (eight significant digits) up to 18th degree polynomials without appreciable round-off error.

In addition, this report contains a complete analysis, in which the general parameter was used, of all the data for three steels that were obtained by NASA through the cooperation of Dr. K. Richard of Faberwerke Hoechst in Frankfurt and that were investigated in a long-time cooperative creep program in Germany. Some of the data from the latter investigation are included in this paper.

Finally it is shown by means of a concrete example how the parameter techniques can be applied to creep data to predict long-time creep. For this purpose the data for columbium alloy FS-85, as reported in reference 8, are used.

A complete Fortran IV program, as used on the IBM 7094 computer in making the calculations, is presented in appendix B. This program can be used for the objective analysis of any set of creep-rupture data by the Larson-Miller, Manson-Haferd, or the more general parameter of reference 4.

SYMBOLS

A,B linear transformation coefficients
a,b,c elements of coefficient matrix
D standard deviation

K	degree of freedom
m	degree of polynomial
n	number of data points
P(σ)	creep-rupture parameter
Q	polynomial
q	stress exponent
r	temperature exponent
S	sum of squares of residuals
T	temperature
T _a	temperature intercept
t	time to rupture
t _a	time intercept
u	coefficient of polynomial function
X	scaled log stress
x	log stress
y	log time
y _a	log time intercept
α, β	constants from recurrence relation
σ	stress
τ	$\sigma^q(T - T_a)^r$

Subscripts:

max	maximum
min	minimum

PROCEDURE

General Parameter

The general creep-rupture parameter introduced in reference 4 has the fol-

lowing form

$$P(\sigma) = \frac{\frac{\log t}{\sigma^q} - \log t_a}{(T - T_a)^r} \quad (1)$$

where T_a , $\log t_a$, q , and r are material constants to be determined from the available experimental data. The parameter $P(\sigma)$ is a function of the stress and, when plotted against stress, is referred to as a master curve (fig. 1, p. 9). If $q = 0$ and $r = 1$, the Manson-Haferd parameter is obtained. If $q = 0$, $r = -1$, and $T_a = -460^{\circ}$ F, the Larson-Miller parameter results. If $q = 1$ and $r = 1$, the stress-modified parameter suggested in reference 9 is obtained. Finally, if $q = 0$, equation (1) reduces to the parameter proposed by Manson and Brown (ref. 10).

The object is to find the best values of the constants q , $\log t_a$, T_a , and r so that the master curve best fits the data. To find these values, the method of least squares is used whereby the master curve is represented by a polynomial in the logarithm of the stress, and the best fit is obtained by minimizing the sum of the squares of the deviations (the residuals) of the data from the curve. The calculation procedure will now be described. The details of the derivation are given in appendix A, and a Fortran IV computer program using this method is given in appendix B.

Calculation Procedure

To simplify the notation, the following symbols are introduced:

$$\left. \begin{array}{l} \tau \equiv \sigma^q(T - T_a)^r \\ y \equiv \log t \\ x \equiv \log \sigma \\ y_a \equiv \log t_a \end{array} \right\} \quad (2)$$

Then from equation (1) it follows that

$$y = \sigma^q y_a + \tau Q(x) \quad (3)$$

where in reference 5, $Q(x)$ was represented by a simple polynomial of the form

$$Q(x) = a_0 + a_1 x + a_2 x^2 + \dots + a_m x^m \quad (4)$$

The least-squares equations obtained sometimes led to difficulties as indicated in the INTRODUCTION. These difficulties can be avoided, however, by rewriting equation (4) in terms of polynomials that are orthogonal over the set of data, as defined in appendix A. Thus assume

$$Q(x) = u_1 Q_1(x) + u_2 Q_2(x) + \dots + u_{m+1} Q_{m+1}(x) = \sum_{j=1}^{m+1} u_j Q_j(x) \quad (5)$$

where u_j is an unknown constant, m is the degree of the highest degree polynomial, and $Q_j(x)$ is a polynomial of degree $j - 1$ that satisfies the orthogonality conditions described in appendix A. The use of orthogonal polynomials permits the solution of the least-squares equations directly in closed form, thus the loss of a large number of significant digits is avoided. The method of calculating Q_j will be discussed in appendix A.

If equation (5) is substituted into equation (3), an equation with $m + 5$ unknown constants results for the case of the general parameter. For the case of the linear parameter there are $m + 3$ constants, and for the Larson-Miller parameter there are $m + 2$. It is necessary that the number of data points n always equals or exceeds the number of unknown constants.

The constants are determined so that equation (3) fits the data best in the least-squares sense. To accomplish this, the sum of the squares of the deviations is minimized; that is,

$$S \equiv \sum_{i=1}^n \left[y_i - \sigma_i^q y_a - \tau_i Q(x_i) \right]^2 \quad (6)$$

is made a minimum. Because the equations are nonlinear in some of the unknown constants a trial and error procedure must be used. A set of values is assumed for q , r , and T_a , and the corresponding best values of y_a and u_j are determined. A different set of values for q , r , and T_a is then chosen, and again the best values of y_a and u_j are calculated. Several sets of values of q , r , and T_a are tried, and the values corresponding to the overall best fit are determined. For the case of the linear parameter, only the value of T_a is varied (q is always equal to zero, and r is always equal to 1). For the Larson-Miller parameter, T_a is equal to -460° F, and no trial and error procedure is needed.

As a measure of the fit, the standard deviation D , defined by

$$D = \sqrt{\frac{S}{n - K}} \quad (7)$$

is used, where K equals

$$\left. \begin{array}{ll} m + 5 & \text{general parameter} \\ m + 3 & \text{linear parameter} \\ m + 2 & \text{Larson-Miller parameter} \end{array} \right\} \quad (8)$$

The smallest value of D will correspond to the best fit.

To determine the best values of y_a and u_j for a given set of values of T_a , q , and r , the following calculations are made. First, the logarithms of the stresses are scaled so that they lie in the range -2 to 2, as suggested in reference 7. The reason for this is discussed in appendix A. Thus define a variable X by

$$X = Ax + B \quad (9a)$$

$$\left. \begin{aligned} A &= \frac{4}{x_{\max} - x_{\min}} \\ B &= -2 \frac{x_{\max} + x_{\min}}{x_{\max} - x_{\min}} \end{aligned} \right\} \quad (9b)$$

The polynomials $Q_j(X_i)$ are now calculated for each of the data points by using the following formulas:

$$Q_{j+1} = (X - \alpha_j)Q_j - \beta_j Q_{j-1} \quad m \geq j \geq 1 \quad (10)$$

$$\left. \begin{aligned} \alpha_j &= \frac{\sum_{i=1}^n x_i \tau_i^2 Q_j^2(x_i)}{\sum_{i=1}^n \tau_i^2 Q_j^2(x_i)} \quad m \geq j \geq 1 \\ \beta_j &= \frac{\sum_{i=1}^n x_i \tau_i^2 Q_j(x_i) Q_{j-1}(x_i)}{\sum_{i=1}^n \tau_i^2 Q_{j-1}^2(x_i)} \quad m \geq j > 1, Q_1 = 1, \text{ and } \beta_1 = 0 \end{aligned} \right\} \quad (10a)$$

where n is the number of data points, X_i is the scaled value of \log for the i^{th} data point, and τ_i is equal to $\sigma_i^q(T_i - T_a)^r$ for the i^{th} data point for the chosen values of T_a , q , and r .

It is to be noted that the degree of the polynomial $Q(x)$ of equation (5) can be increased by merely computing the next polynomial in the series Q_{m+2} without having to recompute any of the previous ones. This is one of the advantages of using orthogonal polynomials.

Once the values of Q_j have been computed for each of the data points, y_a and u_j can be calculated as follows:

Let

$$\left. \begin{aligned}
 a_0 &= \sum_{i=1}^n \sigma_i^{2q} \\
 a_j &= \sum_{i=1}^n \sigma_i^q \tau_i Q_j(x_i) \\
 b_j &= \sum_{i=1}^n \tau_i^2 Q_j^2(x_i) \\
 c_0 &= \sum_{i=1}^n \sigma_i^q y_i \\
 c_j &= \sum_{i=1}^n \tau_i y_i Q_j(x_i)
 \end{aligned} \right\} \quad (11)$$

where $j = 1, 2, \dots, m + 1$.

Then

$$\left. \begin{aligned}
 y_a &= \frac{c_0 - \sum_{j=1}^{m+1} \frac{a_j c_j}{b_j}}{a_0 - \sum_{j=1}^{m+1} \frac{a_j^2}{b_j}} \\
 u_j &= \frac{c_j - a_j y_a}{b_j}
 \end{aligned} \right\} \quad (12)$$

Note that if $q = 0$, a_0 equals the number of data points n . Thus by means of equations (9) to (12), the best values of y_a and u_j to fit the data are found for a given choice of T_a , q , and r . The Fortran IV program described in appendix B automatically scans all the desired values of T_a , q , and r and chooses the best set from all the submitted values as determined by the smallest value of the standard deviation D , as defined by equation (7). The method can be illustrated by a simple example: consider a set of theoretical data, which fit the following equation exactly

$$\frac{9.5 - \log t}{T - 600} = 10^{-3}(7.02 + 0.467 x + 0.061 x^2 + 0.00928 x^3) \quad (13)$$

Eight data points satisfying this equation are given in columns 2 to 6 of table I. For this data $T_a = 600^\circ F$ and $\log t_a = y_a = 9.5$. Suppose, however, that these eight data points were obtained experimentally and that the values of T_a and $\log t_a$ were not known. The problem then is to find the best values of T_a and $\log t_a$ to fit the data by the linear parameter. These values can readily be found by using the equations of the previous section. First, from column 6 of table I

$$(\log \sigma)_{\max} = 4.75051$$

$$(\log \sigma)_{\min} = 1.81954$$

Therefore from equations (9b)

$$A = 1.36474$$

$$B = -4.48319$$

and by means of equation (9a) the X_i were computed and are given in column 8.

For illustrative purposes three values of T_a were chosen, 500° , 600° , and $700^\circ F$. For each of these values of T_a , values of T_i , α_j , β_j , and $Q_j(X_i)$ were computed by means of equations (2), (10), and (10a), and the values of a_j , b_j , and c_j were computed by equations (11). The results are tabulated for $T_a = 600^\circ$ in columns 9 to 12 of table I and in table II up to a third degree polynomial.

The values of y_a and u_j were then computed by using equations (12) for each of these three values of T_a by first assuming $m = 2$, then $m = 3$, and finally $m = 4$, corresponding to polynomials of second, third, and fourth degrees, respectively. For each of these cases the standard deviation D was computed from equation (7) with S being given by equation (6) and Q by equation (5). The results are summarized in table III. The least value of D , signifying the best fit, is obtained for $m = 3$ and $T_a = 600^\circ F$. The corresponding value of y_a is 9.5. These values, of course, correspond to equation (13), from which the data were generated.

Application to Data from German Cooperative Long-Time Creep Program

As part of the German cooperative long-time creep program, a sufficient amount of material of each of three steels was supplied to NASA to permit the running of short-time tests necessary to predict the results at long times obtained in the German test program. The composition of these steels is shown in table IV.

The results of the NASA tests, which were used in the subsequent analysis,

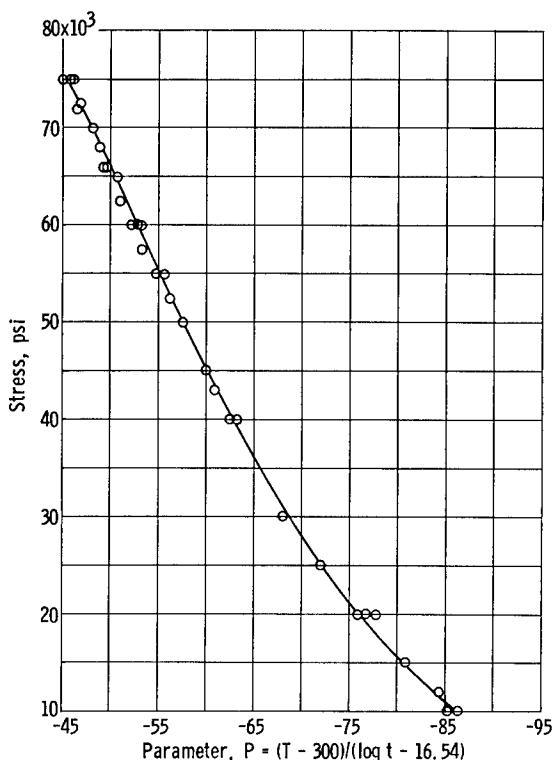


Figure 1. - Master curve for steel K (27b KK), calculated from NASA data between 10 and 3700 hours.

For all three steels the analysis showed the stress exponent q to be zero, but the temperature exponent r to be different for each of the three materials. For steel K the best value of r was 1, which indicated that the best fit is obtained by the linear parameter. For steel P a value of r of -1 was obtained, which indicated a parameter similar to the Larson-Miller parameter; however, the corresponding value of T_a was 200° F rather than -460° F used in the Larson-Miller parameter. For steel C the value of R was 2.5.

Figure 1 shows the results for steel K. Here the master curve consists of a plot of stress against the optimized parameter $(T - 300) / (\log t - 16.54)$.

Figure 2 shows the isothermals computed by using the optimized parameters, as shown on each of the figures. The range of the NASA data used to obtain these parameters is also shown on each of the figures. The data points shown are the German results obtained to date. The predictions up to 100 000 hours from the NASA data based on the optimized parameters agree well with the German data, if scatter and differences in testing technique between the two organizations are considered.

Figure 3 shows a comparison for each of the three steels between the best linear parameter, the best Larson-Miller parameter, and the best general parameter. Although for some of the steels fair agreement can be obtained with one or the other of these parameters, it is clear that the general parameter is superior when all the materials are considered jointly. If any one of the special cases of this parameter is to be chosen for all materials, the linear

are shown in table V. Table VI shows the results of the long-time German test program. The three steels will be designated briefly as steel K, steel C, and steel P.

With the use of the test data shown in table V a complete analysis was made by the previously described method. The general parameter discussed in the INTRODUCTION was used, and the best values were obtained for the parametric constants for each of the three steels.

All the data obtained for these steels are shown in tables V and VI. Many of the data points were obtained for purposes other than the application to time-temperature parameters, as described in this report. As already discussed in references 4 and 11, a much smaller amount of data is needed when an accelerated program is desired; however, since these data were already available, all the data indicated in tables V and VI were used to obtain the best possible parametric constants.

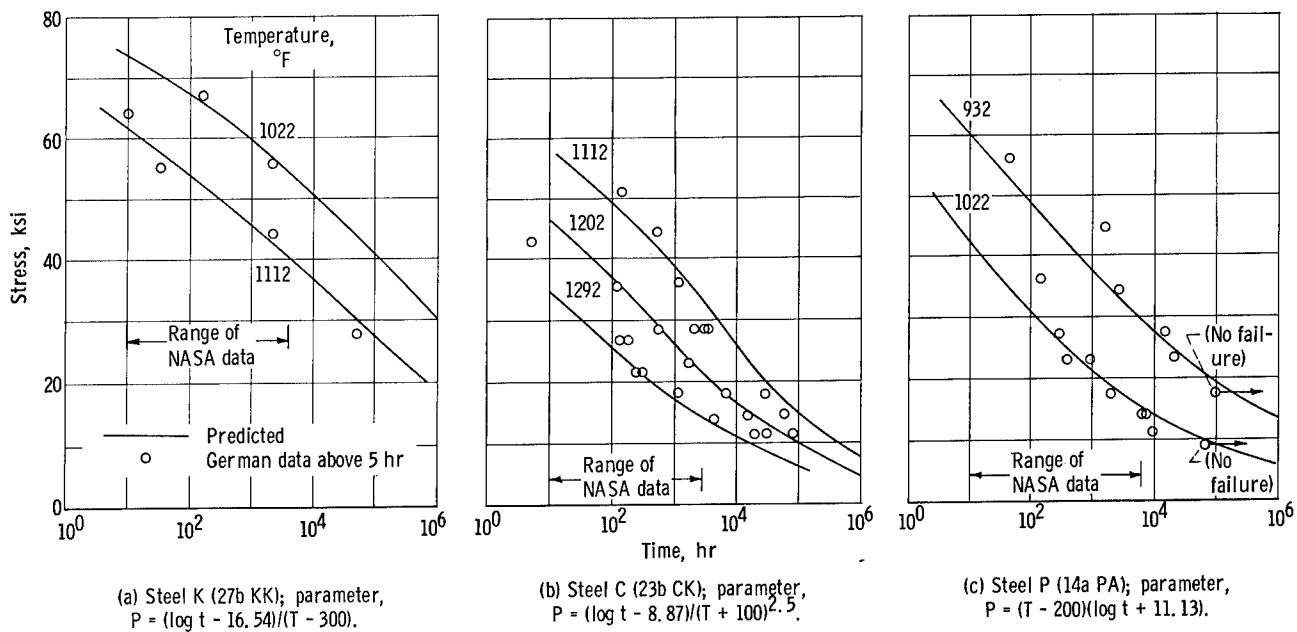


Figure 2. - Analysis of German steel data by generalized parameter with optimum constants (where T is temperature, and t is time to rupture).

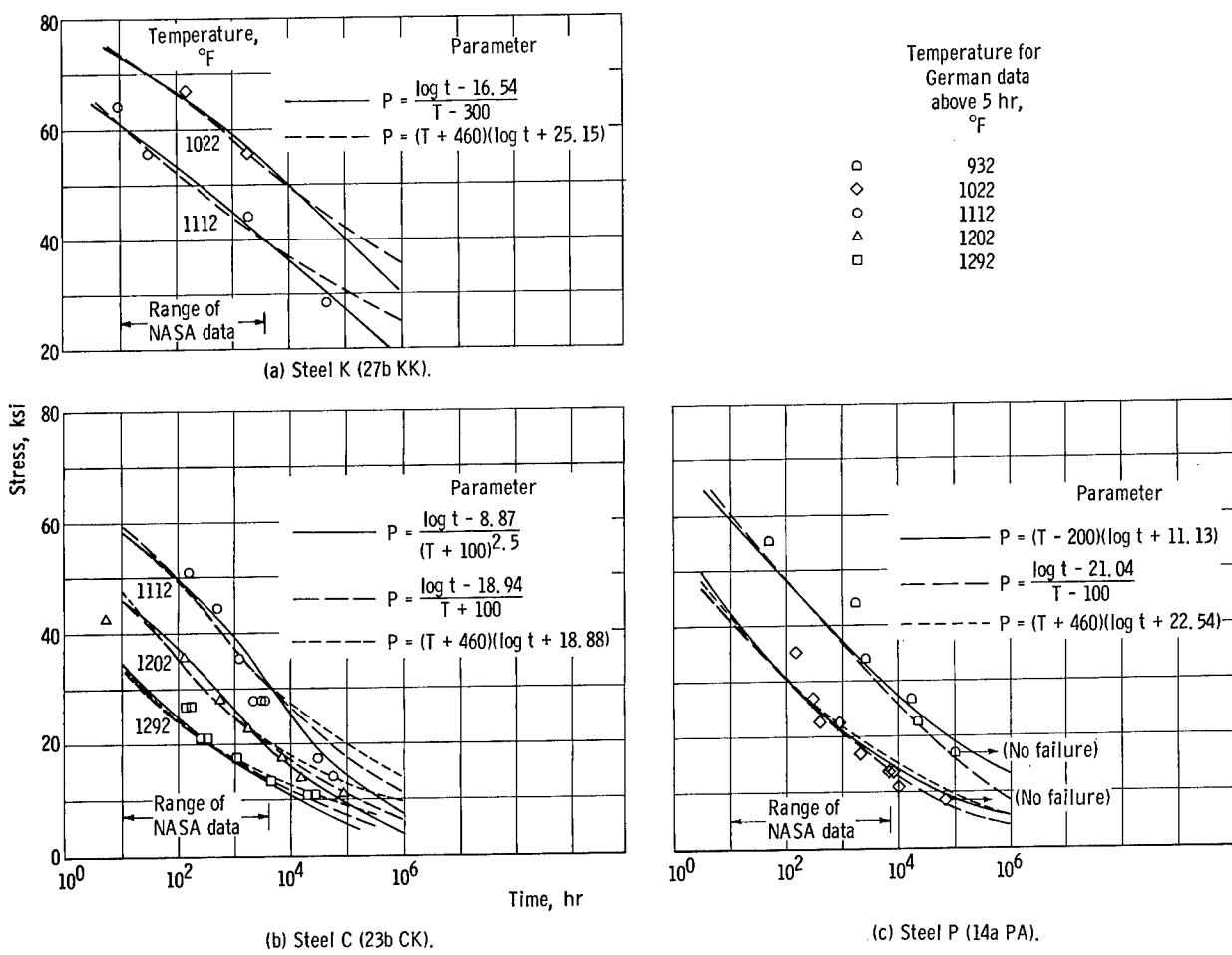
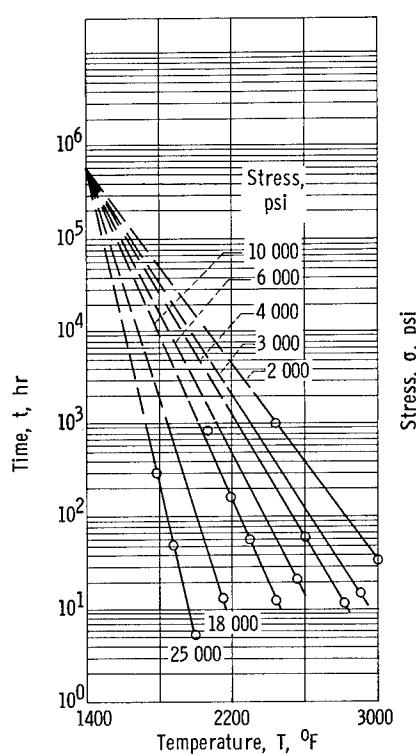
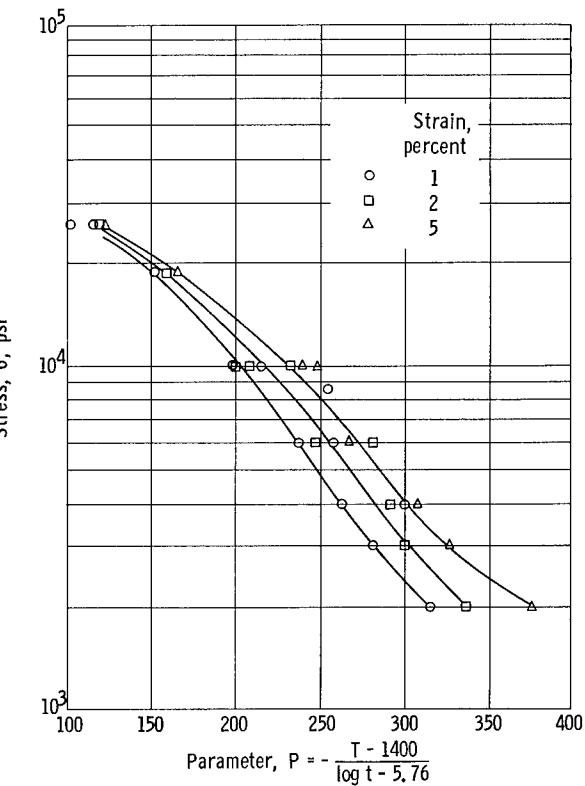


Figure 3. - Analysis of German steel data by several parameters (where T is temperature, and t is time to rupture).



(a) 5-Percent strain.



(b) Master curves obtained for 1-, 2-, and 5-percent strain.

Figure 4. - Analysis of creep data for columbium alloy FS-85 by linear parameter.

parameter would appear to be the best choice.

Application to Creep Data

Although there is no fundamental reason why the same parameter is capable of representing both ~~all~~ creep and rupture data, it has nevertheless been found empirically (refs. 1 and 2) that the dual role of the same parameter leads to reasonable results. Experimental data for creep are much more limited, however, than that for rupture, and such data tend to contain more scatter, hence, analysis of creep data by the parametric approach has been limited in the past.

The method of the present report can be applied directly to creep data without any change. All that is necessary is to redefine t as the time to attain a specified amount of creep rather than as the rupture time. Thus, it is assumed that for a given amount of creep, say 1 percent, a plot of $\log \sigma$ against a parameter, such as that given by equation (1), will produce a single master curve. For a different amount of creep, say 5 percent, a different master curve can be obtained, but it is assumed that the parametric constants, such as $\log t_a$ and T_a , remain the same and that they equal the values obtained from rupture data.

Calculations of this type were performed for columbium alloy FS-85. The creep tests were limited to runs of approximately 1000 hours; the data are

given in table VII, as taken from reference 8. Figure 4(a) shows the data for 5-percent creep strain, and figure 4(b) shows the master curves obtained for 1-, 2-, and 5-percent strain as well as the parametric constants obtained by the method of this report. While scatter in the creep data is high, the correlation must be regarded as good. In general, the points agree well with the master curve.

Although these results are encouraging, much more work is necessary before it can be concluded that the parametric approach is completely valid for creep data. If it is eventually concluded that the parametric approach is valid for creep data and in particular that the parametric constants are the same for both the creep and rupture processes, it is obvious that a great saving in test facilities and test program planning will result. It therefore seems very worthwhile in future studies to give more attention to the correlation and extrapolation of creep data by the parametric method.

Lewis Research Center,
National Aeronautics and Space Administration,
Cleveland, Ohio, May 3, 1965.

APPENDIX A

ORTHOGONAL POLYNOMIALS AND LEAST-SQUARES DETERMINATION OF PARAMETRIC CONSTANTS

A set of polynomials $Q_j(x)$ are said to be orthogonal over an interval with respect to the weighting function $\tau(x)$ if they satisfy the following relation

$$\int_{x=x_1}^{x=x_2} \tau^2(x) Q_j(x) Q_k(x) dx = 0 \quad j \neq k \quad (A1)$$

Similarly a set of polynomials can be defined to be orthogonal over a set of n discrete points x_i by the following relation

$$\sum_{i=1}^n \tau_i^2 Q_j(x_i) Q_k(x_i) = 0 \quad j \neq k \quad (A2)$$

It can be shown (ref. 6), that all orthogonal polynomials satisfy a three-term recurrence relation of the form

$$Q_{k+1} = (x - \alpha_k) Q_k - \beta_k Q_{k-1} \quad k \geq 1 \quad (A3)$$

Thus by starting with $Q_1 = 1$ and $\beta_1 = 0$ an infinite set of orthogonal polynomials can be generated by means of equation (A3) if values for α_k and β_k are known. These can be determined from the orthogonality conditions (eqs. (A1) or (A2)). From the relation (A2) it follows that

$$\sum_{i=1}^n \tau_i^2 Q_k(x_i) Q_{k+1}(x_i) = 0 \quad (A4a)$$

and

$$\sum_{i=1}^n \tau_i^2 Q_{k+1}(x_i) Q_{k-1}(x_i) = 0 \quad (A4b)$$

When the recurrence relation (A3) is used to eliminate Q_{k+1} , there is obtained

$$\sum_{i=1}^n \tau_i^2 Q_k \left[(x_i - \alpha_k) Q_k - \beta_k Q_{k-1} \right] = 0 \quad (A5a)$$

$$\sum_{i=1}^n \tau_i^2 \left[(x_i - \alpha_k) Q_k - \beta_k Q_{k-1} \right] Q_{k-1} = 0 \quad (A5b)$$

When the orthogonality condition (A2) is used, equations (A5a) and (A5b) reduce to

$$\sum_{i=1}^n \tau_i^2 (x_i - \alpha_k) Q_k^2 = 0 \quad (A6a)$$

$$\sum_{i=1}^n \tau_i^2 (x_i Q_k Q_{k-1} - \beta_k Q_{k-1}^2) = 0 \quad (A6b)$$

Solving equations (A6) for α_k and β_k gives

$$\alpha_k = \frac{\sum_{i=1}^n x_i \tau_i^2 Q_k^2}{\sum_{i=1}^n \tau_i^2 Q_k^2} \quad (A7a)$$

$$\beta_k = \frac{\sum_{i=1}^n x_i \tau_i^2 Q_k Q_{k-1}}{\sum_{i=1}^n \tau_i^2 Q_{k-1}^2} \quad (A7b)$$

Thus a set of orthogonal polynomials can be generated that are orthogonal over a finite set of discrete values of the variable x . Note that these values need not be equally spaced, a condition that is obviously necessary for stress-rupture data.

Scaling of Polynomial Argument

From the recurrence relation (A3) with $Q_1 = 1$, it follows that the leading term of $Q_{k+1}(x_i)$ is x_i^k . Therefore, depending on the values of x_i , the values of $Q_{k+1}(x_i)$ can become very large or very small. This procedure can lead to a loss of significant figures in performing the calculations. It is shown in reference 7, by comparison with the Chebyshov polynomials, that if x is scaled so that all the values of x_i lie between 2 and -2, the polynomial

values $Q_j(x_i)$ will all be of approximately uniform size. To perform this scaling, let x_{\max} be the maximum value of $\log \sigma$ and x_{\min} be the minimum value of $\log \sigma$; then let

$$X = A \log \sigma + B \quad (A8)$$

$$2 = Ax_{\max} + B \quad (A9a)$$

$$-2 = Ax_{\min} + B \quad (A9b)$$

and solving for A and B results in equations (9b).

It has been found in practice that scaling the values of x as indicated does indeed preserve the significance of the calculations.

Least-Squares Procedure

In terms of the orthogonal polynomials, equation (3) can be written

$$y = \sigma^q y_a + \tau \sum_{j=1}^{m+1} u_j Q_j(x) \quad (A10)$$

To find the best values of y_a and u_j that fit the data, the sum of the squares of the residuals is minimized. Thus let

$$S = \sum_{i=1}^n \left[y_i - \sigma_i^q y_a - \tau_i \sum_{j=1}^n u_j Q_j(x_i) \right]^2 \quad (A11)$$

Then in order to find the values of y_a and u_j that will make S a minimum, S is differentiated in turn with respect to y_a and each u_j , and the resulting equations are set equal to zero. When this is done, the following set of equations is obtained:

$$\left. \begin{aligned} a_0 y_a + a_1 u_1 + a_2 u_2 + \dots + a_{m+1} u_{m+1} &= c_0 \\ a_1 y_a + b_1 u_1 + 0 + \dots + 0 &= c_1 \\ a_2 y_a + 0 + b_2 u_2 + \dots + 0 &= c_2 \\ \vdots & \quad \vdots \\ \vdots & \quad \vdots \\ a_{m+1} y_a + 0 + 0 + \dots + b_{m+1} u_{m+1} &= c_{m+1} \end{aligned} \right\} \quad (A12)$$

where

$$\left. \begin{aligned}
 a_0 &= \sum_{i=1}^n \sigma_i^{2q} \\
 a_j &= \sum_{i=1}^n \sigma_i^q \tau_i Q_j(x_i) \quad j = 1, 2, \dots, m+1 \\
 b_j &= \sum_{i=1}^n \tau_i^2 Q_j^2(x_i) \quad j = 1, 2, \dots, m+1 \\
 c_0 &= \sum_{i=1}^n \sigma_i^q y_i \\
 c_j &= \sum_{i=1}^n \tau_i y_i Q_j(x_i) \quad j = 1, 2, \dots, m+1
 \end{aligned} \right\} \quad (A13)$$

It is to be noted that the only nonzero elements in the coefficient matrix of equations (A12) are the diagonal elements and the elements of the first row and first column. All the other elements are zero because of the orthogonality properties of the polynomials used. This is one of the major advantages in using orthogonal polynomials. In the usual case of data fitting, all the elements of the first row and first column, except for the first element, would also be zero; and the equations would be completely uncoupled, each u_j being computed completely independent of the others, without the necessity of solving any sets of equations with the resultant loss of significant figures. In this particular case because of the added constant y_a , the equations are not completely uncoupled, but they are very nearly uncoupled and can readily be solved. Thus for any equation after the first

$$u_j = \frac{c_j - a_j y_a}{b_j} \quad (A14)$$

Substituting into the first equation and solving for y_a give immediately

$$y_a = \frac{c_0 - \sum_{j=1}^{m+1} \frac{a_j c_j}{b_j}}{a_0 - \sum_{j=1}^{m+1} \frac{a_j^2}{b_j}} \quad (A15)$$

APPENDIX B

FORTRAN IV PROGRAM

\$ID YAG1202 ERNEST ROBERTS, JR. - 140 M-S - PAX 6132
 \$LIBS10 CONTINUE
 \$IBJOB SOURCE
 \$IBFTC PRMTR1 LIST,REF,DECK
 C CREEP/STRESS-RUPTURE PARAMETER PROGRAM
 C
 C NOMENCLATURE IS AS FOLLOWS
 C
 DD STANDARD DEVIATION PRMT 1
 KK DEGREE OF FREEDOM PRMT 2
 KM NUMBER OF VALUES OF M READ PRMT 3
 KQ NUMBER OF VALUES OF Q READ PRMT 4
 KR NUMBER OF VALUES OF R READ PRMT 5
 KTA NUMBER OF VALUES OF TTA READ PRMT 6
 M DEGREE POLYNOMIAL PRMT 7
 N NUMBER OF DATA POINTS PRMT 8
 PP PARAMETER PRMT 9
 Q STRESS EXPONENT PRMT 10
 QQ POLYNOMIAL PRMT 11
 R TEMPERATURE EXPONENT PRMT 12
 RATIO ABS(Y-YY)/DD PRMT 13
 SIGMA STRESS PRMT 14
 SIGQ SIGMA**Q PRMT 15
 T TIME PRMT 16
 TA TIME INTERCEPT PRMT 17
 TAU SIGMA**Q*(TT-TTA)**R PRMT 18
 TAUSQR TAU**2 PRMT 19
 TIME CALCULATED T (10.**YY) PRMT 20
 TT TEMPERATURE PRMT 21
 TTA TEMPERATURE INTERCEPT PRMT 22
 X LOG SIGMA PRMT 23
 Y LOG T PRMT 24
 YA LOG TA PRMT 25
 YY CALCULATED LOG T PRMT 26
 C
 C ALL QUANTITIES IN COMMON WITH THIS PROGRAM AND THIS PAPER PRMT 27
 C ARE REPRESENTED BY THE SAME SYMBOL, WITH REPEATED PRMT 28
 C LETTERS INDICATING THE UPPER CASE AND GREEK LETTERS BEING SPelled-PRMT 29
 C OUT. PRMT 30
 C
 C PROGRAM EXTRAPOLATES CREEP/STRESS-RUPTURE DATA USING A PRMT 31
 C GENERALIZED PARAMETER PRMT 32
 C PP=(Y/SIGMA**Q-YA)/(TT-TTA)**R, PRMT 33
 C SELECTS PARAMETER PRODUCING SMALLEST RESIDUAL AND OUTPUTS A PRMT 34
 C COMPLETE TABLE. RESULTS OF ALL OTHER VALUES ARE SUMMARIZED IN PRMT 35
 C A SHORTER TABLE. PRMT 36
 C
 C *****INPUT***** PRMT 37
 C
 C TITLE CARD, MODE CARD, AND FIVE (5) SETS OF DATA. AT THE END OF PRMT 38
 C EACH SET OF DATA MUST BE A CARD WITH THE WORD 'END' IN THE FIRST PRMT 39
 C THREE COLUMNS. ALL DATA CARDS (EXCEPTING TITLE AND MODE CARDS) PRMT 40
 C MUST HAVE BLANKS IN THE FIRST THREE COLUMNS. COLUMNS 73-80 ARE PRMT 41
 C IGNORED. PRMT 42
 C
 C TITLE - ANY ALPHAMERIC INFORMATION--HEADS EACH PAGE OF OUTPUT PRMT 43
 C
 C MODE CARD - ONE OF THREE WORDS IN COLUMNS 1-6, 'LARSON', 'LINEAR', PRMT 44
 C OR 'GENRAL'. THIS CARD DEFINES 'KK', THE DEGREE OF PRMT 45
 C FREEDOM, USED IN CALCULATING GOODNESS OF FIT. PRMT 46
 C
 C DATA SET 1--VALUES OF TTA TO BE INVESTIGATED--ONE PER CARD PRMT 47

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C           FORMAT (3X,F10.0)--50 VALUES MAXIMUM           PRMT  59
C
C           DATA SET 2--VALUES OF TEMPERATRE EXPONENT, R, TO BE INVESTIGATED PRMT  61
C                   ONE PER CARD--FORMAT (3X,F10.0)--20 VALUES MAXIMUM           PRMT  62
C                                         PRMT  63
C
C           DATA SET 3--VALUES OF STRESS EXPONENT,Q, TO BE INVESTIGATED PRMT  64
C                   ONE PER CARD--FORMAT (3X,F10.0)--20 VALUES MAXIMUM           PRMT  65
C                                         PRMT  66
C
C           DATA SET 4--DEGREES OF POLYNOMIAL, M, TO BE INVESTIGATED PRMT  67
C                   ONE PER CARD--FORMAT (3X,I2)--MAXIMUM VALUE NOT TO           PRMT  68
C                   EXCEED 20--ZERO MAY NOT BE USED.                         PRMT  69
C                                         PRMT  70
C
C           DATA SET 5--DATA POINTS IN THE ORDER TEMPERATURE, STRESS, AND PRMT  71
C                   TIME--ONE SET PER CARD--FORMAT (3X,3F10.0)                  PRMT  72
C                   THE VALUE OF STRESS IS AUTOMATICALLY DIVIDED BY 1000           PRMT  73
C                   FOR ALL CALCULATIONS EXCEPT FINDING THE LOG STRESS.          PRMT  74
C                   200 SETS MAXIMUM.                                         PRMT  75
C                                         PRMT  76
C
C           *****REPEAT*****
C                                         PRMT  77
C                                         PRMT  78
C
C           EACH OF THE FIVE SETS OF DATA MUST BE FOLLOWED BY A CARD HAVING PRMT  79
C                   THE WORD END IN THE FIRST THREE COLUMNS.                      PRMT  80
C
C           ALL DATA CARDS (EXCEPTING TITLE AND MODE CARDS) MUST HAVE THE PRMT  81
C                   FIRST THREE COLUMNS BLANK.                           PRMT  82
C                                         PRMT  83
C
C           WITHIN EACH SET, DATA MAY BE IN ANY ORDER. IT WILL BE PROCESSED PRMT  84
C                   IN THE ORDER PRESENTED TO THE MACHINE.                      PRMT  85
C                                         PRMT  86
C
C           THE CALCULATIONS ARE PERFORMED IN FOUR (4) LOOPS.                  PRMT  87
C                   GOING FROM INNERMOST TO OUTERMOST, THE QUANTITIES ARE VARIED PRMT  88
C                   IN THE FOLLOWING ORDER
C                   DEGREE POLYNOMIAL, M                                         PRMT  89
C                   VALUE OF TTA                                         PRMT  90
C                   TEMPERATURE EXPONENT, R                           PRMT  91
C                   STRESS EXPONENT, Q                           PRMT  92
C                                         PRMT  93
C                                         PRMT  94
C
C           THE OUTPUT TABLES UTILIZE LESS THAN 120 COLUMNS ON THE PRINTER PRMT  95
C                   AND EXPECT NO CARRIAGE CONTROLS OTHER THAN 1, 0, + AND BLANK. PRMT  96
C
C           A LINE COUNTER IS INCORPORATED TO LIMIT OUTPUT TO 60 LINES PER PRMT  97
C                   PAGE. FOR EACH NEW PAGE THE TITLE AND APPROPRIATE COLUMN HEADINGS PRMT  98
C                   ARE PRINTED. PROGRAM ENDS WITH A TRANSFER TO THE INITIAL READ. PRMT  99
C                                         PRMT 100
C
C           PAGE COUNTING AND ERROR TRAPS MUST BE PROVIDED BY THE OPERATING PRMT 101
C                   SYSTEM.                                         PRMT 102
C                                         PRMT 103
C
C           PROGRAM WITH IBSYS AND IOCSM WILL RUN ON A 16K MACHINE           PRMT 104
C                                         PRMT 105
C                                         PRMT 106
C
C           LOGICAL TRGGR1,TRGGR2,TRGGR3                                     PRMT 107
C
C           DIMENSION TITLE(12),TABLE(6,110),ITBLE(6,110)                   PRMT 108
C                                         PRMT 109
C                                         PRMT 110
C
C           EQUIVALENCE (TABLE(1,1),ITBLE(1,1))                           PRMT 111
C                                         PRMT 112
C
C           COMMON /DATA/SIGMA(201),T(201),TT(201)                         PRMT 113
C
C           1 /TRYSM(21),Q(51),R(51),TTA(51)                                PRMT 114
C
C           2 /FDATA/SIGQ(200),TAU(200),TAUSQR(200),X(200),XX(200),Y(200) PRMT 115
C
C           3 /CALC/PP(200),RATIO(200),TIME(200),YY(200)                   PRMT 116
C
C           4 /END/END/N/N/DD/DD/DEGREE/DEGREE                           PRMT 117
C
C           5 /PLYNML/OT:IER1(4221),YA,OTHER2(63)                         PRMT 118
C                                         PRMT 119
C                                         PRMT 120
C                                         PRMT 121
C
C           INPUT
C
C           1 WRITE (6,9999)
C           READ (5,9001) (TITLE(K),K=1,12)                               PRMT 122
C                                         PRMT 123
C                                         PRMT 124

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      READ (5,9001) DEGREE          PRMT 125
      K = 0                         PRMT 126
10     K = K+1                     PRMT 127
      READ (5,9002) CHECK,TTA(K)   PRMT 128
          IF (CHECK.NE.END) GO TO 10
      KTA = K-1                   PRMT 129
      K = 0                         PRMT 130
15     K = K+1                     PRMT 131
      READ (5,9002) CHECK,R(K)    PRMT 132
          IF (CHECK.NE.END) GO TO 15
      KR = K-1                   PRMT 133
      K = 0                         PRMT 134
20     K = K+1                     PRMT 135
      READ (5,9002) CHECK,Q(K)    PRMT 136
          IF (CHECK.NE.END) GO TO 20
      KQ = K-1                   PRMT 137
      K = 0                         PRMT 138
25     K = K+1                     PRMT 139
      READ (5,9003) CHECK,M(K)    PRMT 140
          IF (CHECK.NE.END) GO TO 25
      KM = K-1                   PRMT 141
      K = 0                         PRMT 142
30     K = K+1                     PRMT 143
      READ (5,9004) CHECK,TT(K),SIGMA(K),T(K)
          IF (CHECK.NE.END) GO TO 30
      N=K-1                       PRMT 144
C
C          END OF INPUT          PRMT 145
C
C          FIND LUG STRESS AND LOG TIME
C
C          DO 100 K=1,N           PRMT 146
      X(K)=ALOG10(SIGMA(K))+3.    PRMT 147
      Y(K)=ALOG10(T(K))          PRMT 148
100    CONTINUE                   PRMT 149
C
C          INITIALIZE CONSTANTS
C
C          DD1=1.E5               PRMT 150
      LINES=51                     PRMT 151
      TRGGR3=.FALSE.              PRMT 152
      NTRY=0                       PRMT 153
C
C          SCALE LOGS OF STRESS
C
C          CALL SCALE             PRMT 154
C
C          FIND HIGHEST DEGREE POLYNOMIAL
C
C          MAX = 0                 PRMT 155
      DO 110 K=1,KM               PRMT 156
      MAX = MAX0(MAX,M(K))       PRMT 157
110    CONTINUE                   PRMT 158
C
C          MAJOR LOOP - CALCULATES ALL Y(A)'S AND RESIDUALS
C          WRITES SUMMARY TABLE
C          FINDS SMALLEST RESIDUAL
C
C          DO 500 K5=1,KQ          PRMT 159
C
C          CALCULATE SIGMA**0
C
C          DO 112 K=1,N           PRMT 160
      SIGQ(K)=SIGMA(K)**Q(K5)     PRMT 161
112    CONTINUE                   PRMT 162
      DO 400 K4=1,KR             PRMT 163

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      DO 300 K3=1,KTA                      PRMT 191
C
C      CALCULATE TAU AND TAU**2            PRMT 192
C
C      DO 120 K=1,N                         PRMT 193
      TDIFF=ABS(TT(K)-TTA(K3))            PRMT 194
      IF (TDIFF) 118,115,118              PRMT 195
115  TAU(K)=0.                            PRMT 196
      GO TO 119                           PRMT 197
118  TAU(K)=SIGQ(K)*TDIFF**R(K4)        PRMT 198
119  TAUSQR(K) = TAU(K)**2              PRMT 199
120  CONTINUE                           PRMT 200
C
C      EVALUATE POLYNOMIALS              PRMT 201
C
      CALL POLY(MAX)                     PRMT 202
C
      DO 200 K2=1,KM                     PRMT 203
C
C      DETERMINE Y(A)                  PRMT 204
C
      CALL YSUBA (M(K2))                PRMT 205
C
C      CALCULATE THEORETICAL LOG TIMES AND TIMES
C
      CALL YTH(M(K2))                  PRMT 206
C
C      COMPUTE RESIDUAL               PRMT 207
C
      CALL RESID(M(K2))                PRMT 208
C
C      MAKE ONE ENTRY IN SUMMARY TABLE
C
      NTRY=NTRY+1                      PRMT 209
      TABLE(1,NTRY)=Q(K5)              PRMT 210
      TABLE(2,NTRY)=R(K4)              PRMT 211
      ITBLE(3,NTRY)=M(K2)             PRMT 212
      TABLE(4,NTRY)=TTA(K3)            PRMT 213
      TABLE(5,NTRY)=YA                PRMT 214
      TABLE(6,NTRY)=DD                PRMT 215
      TRGGR2=NTRY.EQ.2*LINES          PRMT 216
      IF (TRGGR2) GO TO 170           PRMT 217
      GO TO 190                         PRMT 218
C
C      OUTPUTS ONE PAGE OF SUMMARY TABLE
C
C
C      OUTPUT TITLE AND HEADINGS FOR SUMMARY TABLE
C
170  WRITE (6,9005) (TITLE(K),K=1,12),DEGREE
      IF (LINES.EQ.51) WRITE (6,9006) KTA,KR,KQ,KM,N
      WRITE (6,9007)
      TRGGR1=NTRY.LE.LINES
      LIMIT=LINES
      IF (TRGGR1) LIMIT=NTRY
      DO 180 K=1,LIMIT
      WRITE (6,9008) (TABLE(I,K),I=1,2),ITBLE(3,K),(TABLE(I,K),I=4,6)
      IF (TRGGR1) GO TO 180
      KOL2=K+LINES
      IF (TRGGR2) GO TO 175
      IF (KOL2.GT.NTRY) GO TO 180
175  WRITE (6,9009) (TABLE(I,KOL2),I=1,2),ITBLE(3,KOL2),
      1          (TABLE(I,KOL2),I=4,6)
180  CONTINUE
      NTRY=0
      LINES=55

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IF (TRGGR3) GO TO 1000                                PRMT 257
C
C          SAVE VALUES PRODUCING SMALLEST RESIDUAL
C
190      IF (DD1.LE.DD) GO TO 200                                PRMT 258
          M1 = M(K2)                                PRMT 259
          TTA1=TTA(K3)                                PRMT 260
          R1 = R(K4)                                PRMT 261
          Q1 = Q(K5)                                PRMT 262
          YA1 = YA                                PRMT 263
          DD1=DD                                PRMT 264
200      CONTINUE                                PRMT 265
300      CONTINUE                                PRMT 266
400      CONTINUE                                PRMT 267
500      CONTINUE                                PRMT 268
      TRGGR3=.TRUE.                                PRMT 269
      IF (NTRY.NE.0) GO TO 170                                PRMT 270
C
C          END MAJOR LOOP
C
C          OUTPUT OPTIMUM VALUES AND HEADING FOR FULL TABLE
C
1000     CONTINUE                                PRMT 271
1010     WRITE (6,9005) (TITLE(K),K=1,12),DEGREE      PRMT 272
          LINES=3                                PRMT 273
1020     WRITE (6,9010) Q1,R1,M1,TTA1,YA1,DD1      PRMT 274
          LINES=LINES+5                                PRMT 275
1030     WRITE (6,9011)                                PRMT 276
          LINES=LINES+3                                PRMT 277
C
C          CALCULATE THEORETICAL TIMES, RATIOS OF DIFFERENCES
C          TO RESIDUAL, AND VALUES OF THE PARAMETER, FOR THE
C          PARAMETER PRODUCING THE MINIMUM RESIDUAL
C
          DO 1035 K=1,N                                PRMT 278
          TDIFF=ABS(TT(K)-TTA1)                                PRMT 279
          SIGQ(K)=SIGMA(K)**Q1                                PRMT 280
          IF (TDIFF) 1032,1031,1032                                PRMT 281
1031     TAU(K)=0.                                PRMT 282
          GO TO 1034                                PRMT 283
1032     TAU(K)=SIGQ(K)*TDIFF**R1      PRMT 284
1034     TAUSQR(K) = TAU(K)**2      PRMT 285
1035     CONTINUE                                PRMT 286
          DD=DD1                                PRMT 287
          CALL POLY(M1)                                PRMT 288
          CALL YSUBA(M1)                                PRMT 289
          CALL YTH (M1)                                PRMT 290
          CALL RATIO1                                PRMT 291
          CALL PARAM
C
C          OUTPUT FULL TABLE
C
          K = 0                                PRMT 292
1040     K = K+1                                PRMT 293
          WRITE (6,9012) TT(K),SIGMA(K),X(K),T(K),TIME(K),Y(K),YY(K),
          1                                     RATIO(K),PP(K)      PRMT 294
          LINES=LINES+1                                PRMT 295
          IF (K.EQ.N) GO TO 1      PRMT 296
          IF (LINES.LT.60) GO TO 1040      PRMT 297
          WRITE (6,9005) (TITLE(KKK),KKK=1,12),DEGREE      PRMT 298
          WRITE (6,9011)                                PRMT 299
          LINES=6                                PRMT 300
          GO TO 1040                                PRMT 301
C
C          END OF PROGRAM
C

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C		PRMT 323
C	FORMAT STATEMENTS FOR PROGRAM	PRMT 324
C		PRMT 325
C	FORMATS FOR INPUT	PRMT 326
C		PRMT 327
9001	FORMAT (12A6)	PRMT 328
9002	FORMAT (A3,F10.0)	PRMT 329
9003	FORMAT (A3,I2)	PRMT 330
9004	FORMAT (A3,0PF10.0,3PF10.0,0PF10.0)	PRMT 331
C		PRMT 332
C	FORMATS FOR OUTPUT	PRMT 333
C		PRMT 334
C	TITLE (SKIPS TO NEW PAGE)	PRMT 335
C		PRMT 336
9005	FORMAT(1H1,20X,12A6/1H ,30X,A6,10H PARAMETER/1H)	PRMT 337
C		PRMT 338
C	SUMMARY OF INPUT	PRMT 339
C		PRMT 340
9006	FORMAT (1H ,10X,45HCREEP/RUPTURE PARAMETERS ARE INVESTIGATED FOR/11H ,I2,18H VALUE(S) OF T(A),,I3,25H TEMPERATURE EXPONENT(S),,I3,224H STRESS EXPONENT(S), AND,I3,14H POLYNOMIAL(S)/1H ,10X,5HUSING,314,12H DATA POINTS/1H)	PRMT 341
C		PRMT 342
		PRMT 343
		PRMT 344
C		PRMT 345
C	HEADINGS FOR SUMMARY TABLE, ONE LINE OF SUMMARY TABLE	PRMT 346
C		PRMT 347
9007	FORMAT (1H ,2(2X,1HQ,7X,1HR,6X,1HM,5X,4HT(A),5X,4HY(A),4X, 1 8HSTD.DEV.,10X)/1H)	PRMT 348
9008	FORMAT (1H ,0PF5.2,F8.2,15,F9.0,F10.2,1PE11.2)	PRMT 349
9009	FORMAT (1H+,58X,0PF5.2,F8.2,I5,F9.0,F10.2,1PE11.2)	PRMT 350
C		PRMT 351
C	OPTIMUM VALUES	PRMT 352
C		PRMT 353
9010	FORMAT(1H 10X44HVALUES PRODUCING SMALLEST STANDARD DEVIATION/3H0Q=PRMT 355 1F5.2,4H, R=F5.2,4H, M=I2,7H, T(A)=F6.0,7H, Y(A)=F9.3,11H, STD.DEV.PRMT 356 2=1PE9.2/1H0)	PRMT 356
C		PRMT 357
C		PRMT 358
C	HEADINGS FOR FULL TABLE, ONE LINE OF FULL TABLE	PRMT 359
C		PRMT 360
9011	FORMAT (5H TEMP,4X,6HSTRESS,3X,3HLOG,6X,4HTIME,5X,6HCALC1D,5X, 13HLOG,3X,8HCALC LOG,2X6HDEV/SU,3X,9HPARAMETER/1H ,8X,6H(*E-3),2X, 26HSTRESS,14X,4HTIME,5X,4HTIME,4X,4HTIME/1H)	PRMT 361
9012	FORMAT (1H ,0PF5.0,F8.1,F8.3,2F10.1,3F9.3,1PE12.3)	PRMT 362
C		PRMT 363
9999	FORMAT (1H1)	PRMT 364
C		PRMT 365
	END	PRMT 366
		PRMT 367
		PRMT 368

```

$IBFIC PRMBLK LIST,REF,DECK
C      SETS FIRST POLYNOMIAL TO UNITY AT ALL STATIONS AND STORES      PRMB  1
C      ALPHAMERIC CODE WORDS      PRMB  2
C
C      BLOCK DATA      PRMB  3
COMMON /PLYNML/QQ(21,200),OTHERS(85)/END/END/NAMES/NAMES(2)      PRMB  4
      DATA (QQ(1,K),K=1,200)/200*1./,END/3HEND/,      PRMB  5
      1      (NAMES(K),K=1,2)/12HLARSONLINEAR/      PRMB  6
      END      PRMB  7
                                         PRMB  8

$IBFTC PARAM LIST,REF,DECK
C      SUBROUTINE FOR EVALUATING THE PARAMETER AT EACH POINT      PARM  1
C
C      SUBROUTINE PARAM      PARM  2
C
COMMON /FDATA/SIGQ(200),TAU(200),OTHERS(600),Y(200)      PARM  3
      1      /CALC/PP(200),OTHER1(600)/N/N      PARM  4
      2      /PLYNML/OTHER2(4221),YA,OTHER3(63)      PARM  5
C
      DO 10  K=1,N      PARM  6
      PP(K) = (Y(K)-SIGQ(K)*YA)/TAU(K)      PARM  7
10      CONTINUE      PARM  8
      RETURN      PARM  9
      END      PARM 10
                                         PARM 11
                                         PARM 12
                                         PARM 13

$IBFTC YTH LIST,REF,DECK
C      SUBROUTINE FOR CALCULATING TIMES AND LOG TIMES FROM THE PARAMETER YTH  1
C
C      SUBROUTINE YTH(M)      YTH  2
C
COMMON /CALC/OTHERS(400),TIME(200),YY(200)      YTH  3
      1      /FDATA/SIGQ(200),TAU(200),OTHER1(800)      YTH  4
      2      /PLYNML/QQ(21,200),U(21),YA,OTHER2(63)      YTH  5
      3      /N/N      YTH  6
C
      DO 10  K=1,N      YTH  7
      YY(K) = 0.      YTH  8
10      CONTINUE      YTH  9
      M1 = M+1      YTH 10
      DO 30  K=1,N      YTH 11
      DO 20  J=1,M1      YTH 12
      YY(K) = YY(K)+QQ(J,K)*U(J)      YTH 13
20      CONTINUE      YTH 14
      YY(K) = TAU(K)*YY(K)+SIGQ(K)*YA      YTH 15
      TIME(K) = 10.**YY(K)      YTH 16
30      CONTINUE      YTH 17
      RETURN      YTH 18
      END      YTH 19
                                         YTH 20
                                         YTH 21
                                         YTH 22

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$IBFTC RATIO1 LIST,REF,DECK
C      SUBROUTINE FOR CALCULATING RATIOS      RATO  1
C      OF INDIVIDUAL RESIDUALS TO ROOT-MEAN-SQUARE RESIDUAL  RATO  2
C
C      SUBROUTINE RATIO1      RATO  3
C
C      COMMON /FDATA/OTHERS(1000),Y(200)      RATO  4
1      /CALC/OTHER1(200),RATIO(200),OTHER2(200),YY(200)      RATO  5
2      /N/N/DD/DD      RATO  6
C
C      DO 10  K=1,N      RATO  7
10     RATIO(K) = ABS(Y(K)-YY(K))/DD      RATO  8
      CONTINUE      RATO  9
      RETURN      RATO 10
      END      RATO 11
      RATO 12
      RATO 13
      RATO 14

$IBFTC RESID  LIST,REF,DECK
C      SUBROUTINE FOR CALCULATING RESIDUAL      RESD  1
C
C      THE RESIDUAL IS BASED ON THE LOG OF THE TIME.      RESD  2
C      IT IS DEFINED AS THE SQUARE ROOT OF THE SUM OF THE SQUARES OF      RESD  3
C      THE INDIVIDUAL RESIDUALS DIVIDED BY THE DIFFERENCE BETWEEN THE NUMRESD  4
C      BER OF DATA POINTS AND THE DEGREES OF FREEDOM. THE DEGREES OF      RESD  5
C      FREEDOM, KK, DEPENDS ON THE PARAMETER (SEE MAIN BODY OF REPORT).      RESD  6
C          KK=2 FOR LARSON-MILLER PARAMETER      RESD  7
C          KK=3 FOR LINEAR PARAMETER      RESD  8
C          KK=5 FOR GENERAL PARAMETER      RESD  9
C
C      DD = SQRT((Y-YY)**2/(N-M-KK))      RESD 10
C
C      SUBROUTINE RESID(M)      RESD 11
C
C      COMMON /FDATA/OTHERS(1000),Y(200)      RESD 12
1      /CALC/OTHER1(600),YY(200)      RESD 13
2      /DD/DD/N/N/DEGREE/DEGREE/NAMES/FAMES(2)      RESD 14
C
C          IF (DEGREE.EQ.FAMES(2)) GO TO 20      RESD 15
C          IF (DEGREE.EQ.FAMES(1)) GO TO 10      RESD 16
      KK = 5      RESD 17
      GO TO 30      RESD 18
10     KK = 2      RESD 19
      GO TO 30      RESD 20
20     KK = 3      RESD 21
30     D = N-M-KK      RESD 22
      DD = 0.      RESD 23
      DO 40  K=1,N      RESD 24
40     DD = DD+(Y(K)-YY(K))**2      RESD 25
      CONTINUE      RESD 26
      DD = SQRT(DD/D)      RESD 27
      RETURN      RESD 28
      END      RESD 29
      RESD 30
      RESD 31
      RESD 32
      RESD 33
      RESD 34

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$IBFTC YSUBA LIST,REF,DECK
C      SUBROUTINE FOR EVALUATING Y(A)
C
C      THIS SUBROUTINE ALSO EVALUATES THE QUANTITIES, U, NECESSARY
C      FOR DETERMINING THE THEORETICAL LOG TIMES.
C
C      SUBROUTINE YSUBA(M)
C
C      COMMON /PLYNML/QQ(21,200),U(21),YA,A(21),B(21),C(21)
C      1      /FDATA/SIGQ(200),TAU(200),TAUSQR(200),OTHERS(400),
C      2      Y(200)/N/N
C
C      A0 = 0.
C      CO = 0.
C          DO 10  K=1,N
C      A0 = A0+SIGQ(K)**2
C      CO = CO+SIGQ(K)*Y(K)
10    CONTINUE
      M1 = M+1
          DO 20  J=1,M1
      A(J) = 0.
      B(J) = 0.
      C(J) = 0.
20    CONTINUE
          DO 40  J=1,M1
          DO 30  K=1,N
      A(J) = A(J)+SIGQ(K)*TAU(K)*QQ(J,K)
      B(J) = B(J)+TAUSQR(K)*QQ(J,K)**2
      C(J) = C(J) + TAU(K)*Y(K)*QQ(J,K)
30    CONTINUE
40    CONTINUE
      SUM1 = 0.
      SUM2 = 0.
          DO 50  J=1,M1
      AOB = A(J)/B(J)
      SUM1 = SUM1+AOB*C(J)
      SUM2 = SUM2+AOB*A(J)
50    CONTINUE
      YA =(CO-SUM1)/(AO-SUM2)
          DO 60  J=1,M1
      U(J) = (C(J)-A(J)*YA)/B(J)
60    CONTINUE
      RETURN
      END

```

```

$1BFTC POLY      LIST,REF,DECK
C      SUBROUTINE FOR EVALUATING ORTHOGONAL POLYNOMIALS
C
C      ALL POLYNOMIALS UP TO MAXIMUM DESIRED DEGREE ARE EVALUATED
C      AT EACH DATA POINT
C
C      THE FIRST POLYNOMIAL IS IDENTICALLY EQUAL TO UNITY
C      THESE VALUES ARE STORED BY A BLOCK DATA SUBROUTINE
C
C      SUBROUTINE POLY(M)
C
COMMON /FDATA/OTHER1(400),TAUSQR(200),OTHER2(200),XX(200),
1          OTHER3(200)
2          /PLYNML/QQ(21,200),OTHERS(45),ALPHA(20),BETA(20)
3          /N/N
C
S1 = 0.
S2 = 0.
DO 10 K=1,N
S1 = S1+XX(K)*TAUSQR(K)
S2 = S2+TAUSQR(K)
10  CONTINUE
ALPHA(1) = S1/S2
DO 20 K=1,N
QQ(2,K) = XX(K)-ALPHA(1)
20  CONTINUE
IF (M.LE.1) RETURN
DO 50 K=2,M
S1 = 0.
S2 = 0.
S3 = 0.
S4 = 0.
DO 30 J=1,N
D1 = TAUSQR(J)*QQ(K,J)
D2 = D1*QQ(K,J)
S1 = S1+XX(J)*D2
S2 = S2+D2
S3 = S3+XX(J)*D1*QQ(K-1,J)
S4 = S4+TAUSQR(J)*QQ(K-1,J)**2
30  CONTINUE
ALPHA(K) = S1/S2
BETA(K) = S3/S4
DO 40 J=1,N
QQ(K+1,J) = (XX(J)-ALPHA(K))*QQ(K,J)-BETA(K)*QQ(K-1,J)
40  CONTINUE
50  CONTINUE
RETURN
END

```

```

$IBFTC SCALE LIST,REF,DECK
C      SUBROUTINE FOR SCALING LOGS OF STRESS
C
C      THE SCALED VALUES LIE IN THE REGION -2 TO 2
C
C      SUBROUTINE SCALE
C
C      COMMON /FDATA/OTHER1(600),X(200),XX(200),OTHER2(200)/N/N
C
C      BIG = 0.
C      SMALL = 1.E5
C          DO 10 K=1,N
C      BIG = AMAX1(BIG,X(K))
C      SMALL = AMIN1(SMALL,X(K))
10    CONTINUE
C      A = 4.0/(BIG-SMALL)
C      B=2.* (BIG+SMALL)/(BIG-SMALL)
C          DO 20 K=1,N
C      XX(K) = A*X(K)-B
20    CONTINUE
C      RETURN
C      END
C
C      SCAL  1
C      SCAL  2
C      SCAL  3
C      SCAL  4
C      SCAL  5
C      SCAL  6
C      SCAL  7
C      SCAL  8
C      SCAL  9
C      SCAL 10
C      SCAL 11
C      SCAL 12
C      SCAL 13
C      SCAL 14
C      SCAL 15
C      SCAL 16
C      SCAL 17
C      SCAL 18
C      SCAL 19
C      SCAL 20
C      SCAL 21

```

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TABLE I. - CALCULATION OF POLYNOMIALS FOR THEORETICAL DATA FOR THIRD DEGREE POLYNOMIAL

[Temperature intercept, T_a , 600° F.]

1	2	3	4	5	6	7	8	9	10	11	12
Index, i	Tempera- ture, T , °F	Time, t, hr	Stress, σ , psi	log t	log σ	$c^q(T-T_a)^r$	Scaled log σ , X	Polynomial			
								Q_1	Q_2	Q_3	Q_4
1	1100	4954.68	56 300	3.69501	4.75051	500	2.0	1	1.3619	-0.19594	-0.50845
2	1100	11365.9	19 800	4.05560	4.29666	500	1.3806	1	.30341	-1.6875	-.57205
3	1200	625.342	30 300	2.79612	4.48144	600	1.6328	1	.85576	-1.4583	1.7195
4	1200	2908.	5 080	3.46359	3.70586	600	.57433	1	-.80869	1.5741	-1.2980
5	1300	117.371	12 900	2.06956	4.11059	700	1.1267	1	.18925	2.5067	-1.0745
6	1300	1340.	778	3.12710	2.89098	700	-.53777	1	-2.2709	-.90880	.92564
7	1400	34.4856	4 190	1.53764	3.62221	800	.46017	1	2.8030	.015445	-.77354
8	1400	995.25	66	2.99793	1.81954	800	-2.0	1	.51879	-.78251	-.98133

TABLE II. - INTERMEDIATE CALCULATIONS FOR THEORETICAL

DATA FOR THIRD DEGREE POLYNOMIAL

[Temperature intercept, T_a , 600° F.]

1	2	3	4	5	6	7
Index, j	α	β	a	b	c	u
0	-----	-----	8.0	-----	23.743	-----
1	0.27092	0.	5200.	3.48×10^6	14897.	-9.9146×10^{-3}
2	-.57813	1.6548	786.18	5.7589	2616.	-8.4266×10^{-4}
3	.28432	1.3315	465.68	7.6678	3796.9	-8.1771×10^{-5}
4	.41260	.80618	286.01	6.1816	2694.6	-3.6513×10^{-6}

TABLE III. - FIT FOR SEVERAL VALUES OF LINEAR
PARAMETER FOR THEORETICAL DATA

Degree of polynomial	Temperature, T_a	Variable, y_a	Deviation
2	500	10.54	0.008049
3	500	10.54	.009786
4	500	10.55	.010660
2	600	9.49	.004859
3	600	9.50	.000002
4	600	9.50	.000003
2	700	8.44	.015412
3	700	8.46	.013937
4	700	8.45	.014469

TABLE IV. - COMPOSITION OF STEELS RECEIVED
FROM GERMAN COOPERATIVE LONG-
TIME CREEP PROGRAM
[As-received, 20-mm-diam. bar stock.]

Element	Composition, percent		
	Steel		
	C (23b CK)	P (14a PA)	K (27b KK)
Carbon	0.065	0.270	0.068
Silicon	.47	.26	.45
Manganese	.60	.60	.73
Chromium	17.24	2.62	16.14
Molybdenum	2.08	.27	2.10
Columbium and tantalum	.02	Trace	.44
Nickel	11.90	.14	13.12
Titanium	.39	Trace	Trace
Vanadium	.10	.26	.05
Tungsten	Less than 0.005	Trace	Trace

TABLE V. - NASA RUPTURE DATA

(a) Steel K (27b KK)

Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr
1022.00	77 000.000	1.500	1600.00	20 000.000	0.400	^a 1112.00	60 000.000	12.900
^a 1022.00	72 500.000	13.800	1560.00	20 000.000	1.900	^a 1110.00	60 000.000	34.
^a 1022.00	72 000.000	10.	1520.00	20 000.000	4.450	^a 1080.00	60 000.000	52.200
^a 1022.00	70 000.000	36.700	^a 1480.00	20 000.000	23.700	^a 1080.00	60 000.000	37.400
^a 1022.00	68 000.000	60.400	^a 1460.00	20 000.000	25.500	^a 1050.00	60 000.000	239.
^a 1022.00	66 000.000	73.300	^a 1440.00	20 000.000	38.	^a 1030.00	60 000.000	445.
^a 1022.00	66 000.000	107.600	^a 1400.00	20 000.000	136.800	^a 1022.00	60 000.000	989.900
^a 1022.00	65 000.000	201.300	^a 1360.00	20 000.000	394.800	^a 1020.00	60 000.000	817.500
^a 1022.00	62 500.000	250.400	^a 1340.00	20 000.000	704.600	1040.00	75 000.000	.330
^a 1022.00	60 000.000	990.	^a 1320.00	20 000.000	1212.	1022.00	75 000.000	5.850
^a 1022.00	60 000.000	817.500	1320.00	40 000.000	2.700	^a 1000.00	75 000.000	15.600
^a 1022.00	55 000.000	3 680.	1290.00	40 000.000	7.500	^a 980.00	75 000.000	46.500
1112.00	68 000.000	.750	^a 1260.00	40 000.000	15.200	^a 960.00	75 000.000	138.
1112.00	65 000.000	2.250	^a 1230.00	40 000.000	44.400	^a 940.00	75 000.000	542.
1112.00	62 500.000	4.300	^a 1170.00	40 000.000	377.	^a 920.00	75 000.000	579.600
^a 1112.00	60 000.000	13.900	^a 1140.00	40 000.000	1417.	^a 1120.00	50 000.000	186.100
^a 1112.00	57 500.000	22.700	^a 1125.00	40 000.000	2110.	^a 1200.00	40 000.000	130.200
^a 1112.00	55 000.000	51.500	^a 1112.00	40 000.000	5367.	^a 1280.00	30 000.000	132.700
^a 1112.00	52 500.000	147.500	1200.00	60 000.000	.610	^a 1340.00	25 000.000	125.800
^a 1112.00	50 000.000	283.	1170.00	60 000.000	1.250	^a 1500.00	15 000.000	51.300
^a 1112.00	45 000.000	1 020.	1150.00	60 000.000	4.400	^a 1560.00	12 000.000	41.700
^a 1112.00	43 000.000	1 579.	1140.00	60 000.000	4.500	^a 1580.00	10 000.000	32.400
^a 1112.00	37 000.000	13 140.	^a 1120.00	60 000.000	10.900	^a 1540.00	10 000.000	148.200

(b) Steel C (23b CK)

Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr
^a 1600.00	5 000.000	570.200	^a 1230.00	30 000.000	175.700	^a 1202.00	36 000.000	68.600
^a 1620.00	5 000.000	186.600	^a 1250.00	30 000.000	103.500	^a 1202.00	38 000.000	59.300
^a 1660.00	5 000.000	156.800	^a 1280.00	30 000.000	58.100	^a 1202.00	42 000.000	24.400
^a 1680.00	5 000.000	91.600	1292.00	30 000.000	21.300.	^a 1202.00	44 000.000	14.500
^a 1700.00	5 000.000	62.700	^a 1310.00	30 000.000	22.500	^a 1202.00	45 000.000	22.900
^a 1740.00	5 000.000	40.500	^a 1112.00	40 000.000	667.900	1202.00	46 000.000	7.
^a 1780.00	5 000.000	10.600	^a 1120.00	40 000.000	785.400	1202.00	48 000.000	2.850
^a 1425.00	10 000.000	1690.	^a 1150.00	40 000.000	266.700	1202.00	49 000.000	2.550
^a 1450.00	10 000.000	550.300	^a 1170.00	40 000.000	127.800	1202.00	50 000.000	1.470
^a 1480.00	10 000.000	270.	^a 1202.00	40 000.000	44.100	^a 1292.00	18 000.000	859.700
^a 1500.00	10 000.000	170.	^a 1202.00	40 000.000	74.	^a 1292.00	23 000.000	194.600
^a 1520.00	10 000.000	128.500	^a 1210.00	40 000.000	40.500	^a 1292.00	25 000.000	75.
^a 1560.00	10 000.000	40.	^a 1220.00	40 000.000	37.800	^a 1292.00	28 000.000	34.600
^a 1570.00	10 000.000	31.500	^a 1240.00	40 000.000	17.200	^a 1292.00	29 000.000	31.
^a 1600.00	10 000.000	15.800	1270.00	40 000.000	4.500	^a 1292.00	32 000.000	13.300
1650.00	10 000.000	5.250	1280.00	40 000.000	1.200	^a 1292.00	33 000.000	19.800
1700.00	10 000.000	1.750	1292.00	40 000.000	1.300	^a 1292.00	34 000.000	10.400
^a 1202.00	20 000.000	3307.	1300.00	40 000.000	.800	1292.00	36 000.000	2.750
^a 1260.00	20 000.000	667.400	^a 1112.00	34 000.000	2 274.	1292.00	37 000.000	7.600
^a 1290.00	20 000.000	255.	^a 1112.00	43 000.000	363.100	1292.00	38 000.000	1.650
^a 1292.00	20 000.000	347.100	^a 1112.00	46 000.000	233.900	^a 1060.00	60 000.000	42.500
^a 1292.00	20 000.000	363.	^a 1112.00	46 000.000	261.400	^a 1300.00	25 000.000	89.600
^a 1320.00	20 000.000	180.400	^a 1112.00	48 000.000	183.100	^a 1360.00	19 000.000	95.
^a 1360.00	20 000.000	82.	^a 1112.00	50 000.000	84.500	^a 1430.00	15 000.000	71.400
^a 1400.00	20 000.000	28.900	^a 1112.00	52 000.000	65.600	^a 1480.00	12 000.000	147.900
1440.00	20 000.000	9.	^a 1112.00	54 000.000	39.300	^a 1570.00	8 000.000	104.
1480.00	20 000.000	2.500	^a 1112.00	57 000.000	23.300	^a 1630.00	6 000.000	140.900
^a 1112.00	30 000.000	4258.	^a 1202.00	25 000.000	1 074.	^a 1140.00	34 000.000	1077.
^a 1160.00	30 000.000	1110.	^a 1202.00	34 000.000	199.400	^a 1320.00	15 000.000	1505.
^a 1180.00	30 000.000	696.300	^a 1202.00	35 000.000	124.300	^a 1480.00	8 000.000	2237.
^a 1202.00	30 000.000	350.				^a 1540.00	6 000.000	1258.

^aData point used in parametric analysis.

TABLE V. - Concluded. NASA RUPTURE DATA

(c) Steel P (14a PA)

Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr
932.00	65 000.000	3.800	a1250.00	10 000.000	19.200	a740.00	90 000.000	57.100
a932.00	60.000.000	14.150	a1220.00	10 000.000	42.	a785.00	80 000.000	84.
a932.00	60 000.000	14.400	a1180.00	10 000.000	167.	a820.00	70 000.000	195.800
a932.00	57 500.000	10.	a1170.00	10 000.000	203.400	a880.00	60 000.000	120.
a932.00	55 000.000	18.900	a1140.00	10 000.000	608.	a932.00	50 000.000	103.500
a932.00	52 500.000	.51.	a1090.00	10 000.000	2639.	a1022.00	30 000.000	186.700
a932.00	40 000.000	623.	1100.00	40 000.000	1.300	a1050.00	25 000.000	123.500
a932.00	30 000.000	7 592.	1080.00	40 000.000	2.200	a1090.00	20 000.000	79.500
932.00	27 000.000	11 410.	1060.00	40 000.000	4.300	a1090.00	20 000.000	112.400
1022.00	58 000.000	.580	1050.00	40 000.000	6.800	a1120.00	16 000.000	183.500
1022.00	55 000.000	.717	a1040.00	40 000.000	7.400	a1160.00	13 000.000	100.300
1022.00	50 000.000	1.280	a1020.00	40 000.000	22.500	a1230.00	8 000.000	97.900
1022.00	47 000.000	2.450	a1010.00	40 000.000	20.100	a1290.00	5 000.000	139.700
1022.00	47 000.000	6.200	a1000.00	40 000.000	63.300	a740.00	80 000.000	996.600
1022.00	45 000.000	3.500	a990.00	40 000.000	51.200	a780.00	70 000.000	1122.
1022.00	42 500.000	6.300	a980.00	40 000.000	80.600	a830.00	60 000.000	948.800
a1022.00	40 000.000	22.500	a960.00	40 000.000	192.100	a880.00	50 000.000	599.
a1022.00	37 500.000	12.	a940.00	40 000.000	427.900	a932.00	35 000.000	1902.
a1022.00	25 000.000	382.200	a930.00	40 000.000	623.	a980.00	30 000.000	754.800
1415.00	10 000.000	.170	a900.00	40 000.000	2572.	a1000.00	25 000.000	970.700
1340.00	10 000.000	1.500	932.00	70 000.000	1.400	a1030.00	20 000.000	1084.
1315.00	10 000.000	3.700	897.00	70 000.000	5.800	a1070.00	16 000.000	804.800
1290.00	10 000.000	6.100	a860.00	70 000.000	31.200	a1150.00	8 000.000	948.500
						a1220.00	5 000.000	960.

^aData point used in parametric analysis.

TABLE VI. - GERMAN RUPTURE DATA

Temperature, T, °F	Stress, σ, psi	Time, t, hr	Temperature, T, °F	Stress, σ, psi	Time, t, hr
Steel K (27b KK)					
1022.00	76 899.999	0.100	1292.00	17 800.000	1 100.
1022.00	66 899.999	160.	1292.00	21 400.000	300.
1022.00	55 500.000	2 000.	1292.00	21 400.000	250.
1112.00	72 500.000	.100	1292.00	27 000.000	180.
1112.00	64 000.000	10.	1292.00	27 000.000	140.
1112.00	55 500.000	35.	1292.00	47 000.000	.100
1112.00	44 100.000	2 100.			
1112.00	28 400.000	52 000.			
Steel C (23b CK)					
1112.00	14 200.000	60 000.	932.00	84 000.000	0.100
1112.00	17 800.000	30 000.	932.00	75 500.000	.100
1112.00	28 400.000	3 500.	932.00	78 399.999	2.
1112.00	28 400.000	3 000.	932.00	55 500.000	150.
1112.00	28 400.000	2 200.	932.00	44 100.000	1 700.
1112.00	35 600.000	1 200.	932.00	34 200.000	2 600.
1112.00	44 100.000	520.	932.00	27 000.000	16 000.
1112.00	51 200.000	150.	1022.00	17 100.000	100 000.
1112.00	59 800.000	.100	1022.00	72 599.999	.100
1202.00	11 400.000	82 790.	1022.00	69 699.999	.100
1202.00	14 200.000	15 000.	1022.00	65 500.000	1.200
1202.00	17 800.000	6 500.	1022.00	59 800.000	1.500
1202.00	22 800.000	1 800.	1022.00	35 600.000	150.
1202.00	28 400.000	550.	1022.00	27 000.000	300.
1202.00	35 600.000	124.	1022.00	22 800.000	400.
1202.00	42 700.000	5.	1022.00	22 800.000	900.
1202.00	52 600.000	.100	1022.00	17 100.000	2 100.
1292.00	11 400.000	30 000.	1022.00	13 900.000	6 500.
1292.00	11 400.000	20 000.	1022.00	13 900.000	8 000.
1292.00	13 900.000	4 500.	1022.00	11 100.000	10 000.
			1022.00	8 830.000	68 000.

TABLE VII. - CREEP DATA FOR COLUMBIUM ALLOY FS-85

Temperature, T, °F	Stress, σ, psi	Time, t, hr		
		1-Percent creep	2-Percent creep	5-Percent creep
2005	25 000	0.6	3.0	6.1
1900	25 000	26.	33.	45.
1790	25 000	210.	257.	332.
2175	18 000	4.9	7.8	13.
2400	10 000	3.4	5.7	10.8
2300	10 000	25.4	41.	68.
2200	10 000	54.	84.	133.
2100	10 000	355.	500.	765.
2100	10 000	380	570.	875.
2000	10 000	775.	1325.	2175.
2000	10 000	900.	1420.	-----
2000	8 500	2480.	-----	-----
2575	6 000	5.6	10.	22.2
2200	6 000	425.	710.	1370.
2800	4 000	3.4	6.4	13.5
2620	4 000	14.4	26.	56.
2200	4 000	1140.	-----	-----
2900	3 000	2.6	5.4	13.8
3000	2 000	4.6	9.5	33.2
2450	2 000	-----	-----	950.

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